## INSTRUCTIONS A-4

for CESSNA Windshield and the ELIMINATION of the CENTER STRIP

<table>
<thead>
<tr>
<th>Cessna Model Numbers</th>
<th>Part # W-, W/T- or W/G- STC Number</th>
<th>Serial Numbers</th>
<th>Weight #</th>
<th>Moment in. #</th>
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</thead>
<tbody>
<tr>
<td>170</td>
<td>2060 SA210GL</td>
<td>18003 thru 18729</td>
<td>+ 3.8</td>
<td>+ 65</td>
</tr>
<tr>
<td>170A, 170B</td>
<td>2055 SA210GL</td>
<td>18728 thru 27169</td>
<td>+ 3.8</td>
<td>+ 65</td>
</tr>
<tr>
<td>172, 172A</td>
<td>2055 SA211GL</td>
<td>28000 thru 17247746</td>
<td>+ 3.8</td>
<td>+ 65</td>
</tr>
<tr>
<td>172B, 172C</td>
<td>2056 SA211GL</td>
<td>17247747 thru 17249544</td>
<td>+ 3.8</td>
<td>+ 65</td>
</tr>
<tr>
<td>175, 175A, 175B, &amp; 175C</td>
<td>2056 SA212GL</td>
<td>55001 thru 17557119</td>
<td>+ 3.8</td>
<td>+ 65</td>
</tr>
<tr>
<td>180</td>
<td>2072 SA471GL</td>
<td>30000 thru 31261</td>
<td>+ 3.8</td>
<td>+ 65</td>
</tr>
<tr>
<td>180, 180A</td>
<td>2072 SA471GL</td>
<td>31261 thru 50355</td>
<td>No change</td>
<td>No change</td>
</tr>
<tr>
<td>180B, 180C, 180D, 180E, 180F, 180G</td>
<td>2073 SA471GL</td>
<td>50356 &amp; up</td>
<td>No change</td>
<td>No change</td>
</tr>
<tr>
<td>182, 182A</td>
<td>2072 SA301GL</td>
<td>33000 thru 51556</td>
<td>No change</td>
<td>No change</td>
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<tr>
<td>182B, 182C, 182D</td>
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<td>51557 thru 18253598</td>
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<td>No change</td>
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<tr>
<td>182E, 182F, 182G</td>
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<td>18253599 thru 18255844</td>
<td>+4.4</td>
<td>+ 75</td>
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<tr>
<td>185, 185A, 185B, 185C, 185D, 185E, A185E, A185F</td>
<td>2073 SA472GL</td>
<td>185-0001 &amp; up</td>
<td>No change</td>
<td>No change</td>
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<tr>
<td>206, &amp; U206, P206</td>
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<td>206-0001 thru U206-0437</td>
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<td>+ 75</td>
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<td>210-5 (205), 210-5A (205A)</td>
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<td>205-0001 &amp; UP</td>
<td>+4.4</td>
<td>+ 75</td>
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<tr>
<td>210B, 210C, 210D, 210E</td>
<td>2065 or 2065-CM SA393GL</td>
<td>21057841 thru 21058715</td>
<td>+4.4</td>
<td>+ 75</td>
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</tbody>
</table>
MATERIALS:
1. 1 Windshield as noted above
2. 4 felt strips (1 pc M-1001-2 46.5", 2 pcs M-1001-1 18.5", 1 pc M-1001-1 76.5")
3. 4 elastomer Bushings (M-1000-1)
4. 4 pcs, AN-526-632R10 screws
5. 4 pcs, AN-364-632 nuts
6. Sealant - optional - not supplied
7. Miscellaneous hardware (rivets, etc.) - not supplied

INSTRUCTIONS (removal of Cessna Windshield Center strip)
1. Remove center strip (both inner and outer)
2. Remove lower fairing and wing root fairings
3. Remove old windshield by sliding the windshield forward out of the top and side channels
4. Clean channels and mounting surfaces that will contact the windshield. Straighten and repair any nicks or distorted metal that will contact the windshield as required.
5. Place the new windshield in position and check the fit of the windshield - trim if required. See trimming instructions.
6. Again place the windshield in position and temporarily attach the lower fairing and wing fairings so that the windshield is held in the exact final position.
7. Remove the protective paper from the double stick tape in the drill template and carefully locate the template on the top of the windshield and over the top of the upper channel, aligning the template with the marks on the windshield. Check the edge distance of the holes to the edge of the skin - it should be at least 1/4 inch. Mark the hole locations carefully. A No. 27 drill may be used to start the hole in the skin and thus leaving a mark. It is important that this step be done very carefully as the alignment of the holes depends on it.
8. Remove the template and the windshield and finish drilling (with a No. 27 drill) the 4 holes thru the outer and inner walls of the top channel as marked. Remove any burrs and chips that the drilling created.
9. Install the 4 bushings in the holes and attach the self-adhesive felt strips in the windshield. Cut small holes thru the felt to allow the screws to go through the 4 bushings.
10. Place the windshield in position and install the 4 screws and nuts. Tighten the screws lightly, they should be only finger tight and should not distort the metal skin. Over-tightening can cause crazing and premature failure of the windshield.
11. Reinstall the lower fairing and wing root fairings. (Note: sealant may be used in addition to the felt to help prevent leakage.)
12. Fill the holes remaining from the center strip attachment points using the screws and nuts (AN-526 and AN 364) or rivets.
13. Touch up or paint as needed.
14. Relocate compass as desired. A C-2400L4P compass may be located on top of the instrument panel or a standard compass in the instrument panel.
14A. For part numbers 2071-CM & 2065-CM reference instructions A-4CM.
15. New Weight and balance information (see above)
16. Fill out 337 form

NOTE: Some models have the outside air temperature gauge through the center strip of the windshield - this can be moved to the wing root air vent as is common in other Cessna models.
The Head of the screw may be sealed with Silicone Rubber, Dichromate Putty or other approved sealant.

A #27 drill may be used to mark the skin by drilling gently and starting the hole but not drilling thru.

The end of the screw may be cut-off flush.
SUGGESTIONS FOR TRIMMING
ACRYLIC WINDSHIELDS AND WINDOWS

1. Acrylic Windshields are tough - BUT THEY ARE NOT UNBREAKABLE - use care in handling and working

2. A Band saw it the most common way to trim the edges of aircraft Windshields. Use a blade with at least 24 teeth per inch and a high speed - about the same speed as is used for wood.

DO NOT use reciprocating saws, such as saber saws!!!

3. A high speed router may be used for trimming, however, a fixture guide or a very steady hand is required to use this method.

4. To remove smaller amounts from the edge, generally 1/8 inch or less, a hand held disk sander may be used. A medium grit paper, and very light pressure on the disk sander produces good results. A belt sander may be used.

5. Hand tools, such as files, hand saws, sandpaper may be used. Be careful, any binding of saws or files may cause breakage.

6. Before installing the Windshield, carefully smooth the edges to remove notches. Acrylics are notch sensitive

7. Any drilling must be done with drills that are especially sharpened for the purpose. See drawing

8. Installation should be done in accordance with aircraft manufactures recommended procedure and/or normally accepted aircraft maintenance practice

NOTE: We do not recommend the use of covers on Windshields as they may cause crazing

For more detailed information you may also reference AC 43-13-1B Chapter 3
This information is available on the FAA's web site or our web site.